

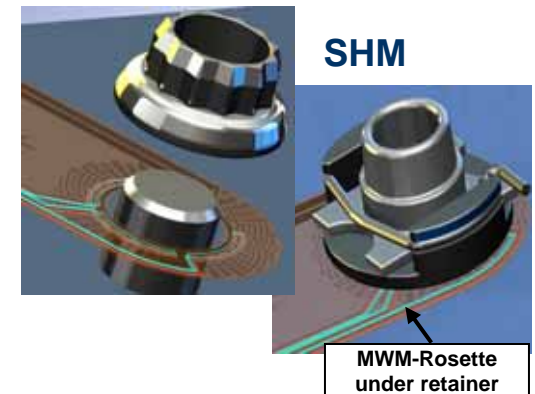
# Calibration vs Calibration Verification for POD Studies & “Reliability”

30 October 2012

Neil Goldfine<sup>1</sup> and Floyd Spencer<sup>2</sup>

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2. Sfhire, Albuquerque, NM 87112-4924; Phone: 505-275-2091; Email: sfhire@comcast.net

JENTEK issued and exclusively licensed patents include U.S. Patent #s 8,222,897, 8,050,883, 7,994,781, 7,876,094, 7,812,601, 7,696,748, 7,589,526, 7,533,575, 7,528,598, 7,526,964, 7,518,360, 7,467,057, 7,451,657, 7,451,639, 7,411,390, 7,385,392, 7,348,771, 7,289,913, 7,280,940, 7,230,421, 7,188,532, 7,183,764, 7,161,351, 7,161,350, 7,106,055, 7,095,224, 7,049,811, 6,995,557, 6,992,482, 6,952,095, 6,798,198, 6,784,662, 6,781,387, 6,727,691, 6,657,429, 6,486,673, 6,433,542, 6,420,867, 6,380,747, 6,377,039, 6,351,120, 6,198,279, 6,188,218, 6,144,206, 5,966,011, 5,793,206, 5,629,621, 5,990,677 and RE39,206 (other US/foreign patents issued and pending).



# Outline

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- **Reliability Related Definitions**
- **Technical Challenges (introduced by Rummel\*) for Calibration Verification**
- **JENTEK Model-Based Calibration, Calibration Verification, and Measurement Approach**
  - For ET NDT
  - For ET SHM(using MWM-Arrays)

\* Rummel, Ward, "Nondestructive Inspection Reliability - History, Status and Future Path", 18th World Conference on Nondestructive Testing, Durban, South Africa, 16-20 April 2010

# Definitions (by Floyd Spencer, with references to Ward Rummel)

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- **Reliability *can be defined as***  
...the probability of a device (or process) performing its defined purpose ***adequately*** for a specified period of time, ***under the operating conditions encountered***
- **NDT Reliability (from Rummel\*)**
  - Reproducibility – Calibration
  - Repeatability – ***Process Control***
  - Capability – POD

\* Rummel, Ward, “Nondestructive Inspection Reliability - History, Status and Future Path”, 18th World Conference on Nondestructive Testing, Durban, South Africa, 16-20 April 2010

# Definitions (by Floyd Spencer) (cont)

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- **Reproducibility (driven by Calibration, according to Ward Rummel)**  
...variability in the device (or process) caused by differences in the behavior of “components” (inspectors/instruments/probes/scanners....)  
  
“Adjustments made to reproduce sensor gain may **change the POD** and off-sets may be necessary” (from Rummel \*) – *in other words do the assumptions for your POD study apply to the inspection?*
- **Repeatability (driven by Process Control, according to Ward Rummel)**  
...variability within fixed “components” due to test – retest
- **Capability (equated to POD, according to Ward Rummel)**  
...measure of the ability of a process to achieve its objectives

# Definitions (by Floyd Spencer) (cont)

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- **Probability of Inspection (POI)**

...probability that field inspection occurs under assumed conditions

*...in other words did your inspection meet the assumptions on which POD is determined*

## Law of total probability as applied to detection for field inspections

Pr(detect) =

Pr(detect | field inspection conditions "A") \* Pr ( field inspection conditions "A") +  
Pr(detect | field inspection conditions "not A") \* Pr (field inspection conditions  
"not A")

$$\text{Pr(detect)} = \text{POD} * \text{POI} + \text{unquantified POD} * (1 - \text{POI})$$

# Definitions (by Floyd Spencer) (cont)

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- **Calibration (or standardization)**  
...comparison of NDT signal response to known flaw characteristics through the use of **reference standards**
- **Calibration in Air or on unflawed parts**  
...for instrument only, requires **verification on reference standards with known flaws**
- **Robustness = Reliability with an emphasis on**  
...a device (or process) performing its defined purpose adequately for a specified period of time, under the operating conditions encountered, **where the operating conditions may vary significantly**

# Calibration Verification

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“If we intend to analyze data by assuming a linear increase in NDT response with increasing crack size, **we are obligated to verify that the measurement system is producing the assumed response.**”\*

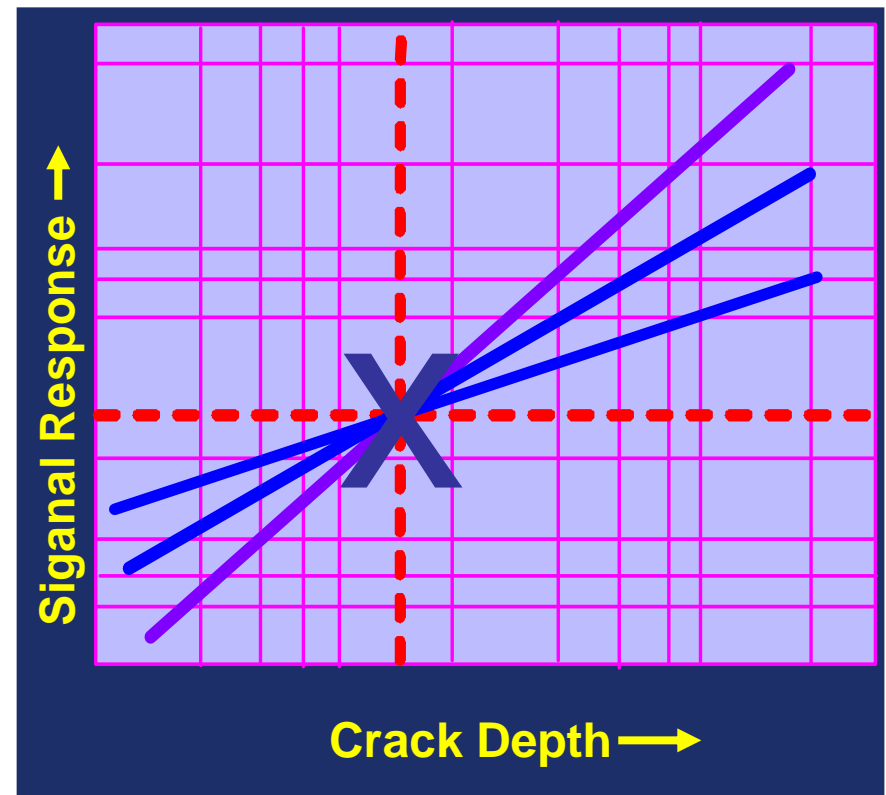
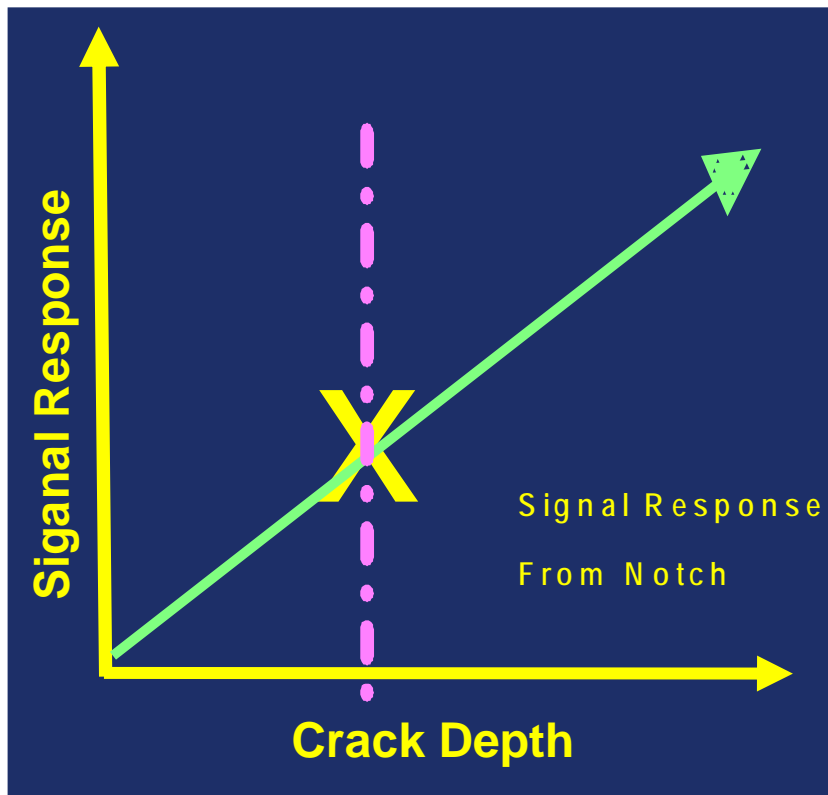
“Instrument calibration should be performed in accordance with manufacturer’s instructions. A permissible instrument calibration is an air standardization with extensive and documented performance verification measurements per manufacturer’s instructions.”\*\*

\* Rummel, Ward, “Nondestructive Inspection Reliability - History, Status and Future Path”, 18th World Conference on Nondestructive Testing, Durban, South Africa, 16-20 April 2010

\*\*E2338 – 04: “Standard Practice for Characterization of Coatings Using Conformable Eddy-Current Sensors without Coating Reference Standards”

# Ward Rummel's suggested approach

“...verify that the measurement system is producing the assumed response,”  
using a three point calibration verification with crack standards



\* Rummel, Ward, "Nondestructive Inspection Reliability - History, Status and Future Path",  
18th World Conference on Nondestructive Testing, Durban, South Africa, 16-20 April 2010



# Calibration Methods (Which is More Robust?)

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Conventional Calibration  
(“Standardization”) on crack standards

*How do we verify that the POD curve (“capability”) we are assuming actually applies to the inspection we are performing?*

*What is a sufficient calibration verification?*

*What if your calibration standards have a different ...?  
(e.g., roughness or paint thickness)*

Model-Based Calibration, with  
Calibration Verification

*Do Model-Based methods provide a more robust solution and means for ensuring that the POD curve assumptions are upheld?*

*What is a sufficient calibration verification?*

*Can we perform statistical process control on our NDT process?  
e.g., monitor parameters that define/constrain the performance*

*The choice is to measure it (verify) or control it (trust)!*

# Technical Challenges (from Ward Rummel)

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“Develop and apply multiple point calibration as a “STANDARD PROCEDURE” – *is this enough?*”

“Link use of predictive NDT performance models to NDT procedure CALIBRATION and NDT acceptance criteria” – *is this better?*”

\* Rummel, Ward, “Nondestructive Inspection Reliability - History, Status and Future Path”, 18th World Conference on Nondestructive Testing, Durban, South Africa, 16-20 April 2010

# JENTEK's Model-Based Approach

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- **Model-Based** Calibration and Measurement Methods
- Calibration Verification on Crack Standards or actual service hardware with verified defects, if available
- **Calibration verification at each inspection location**, to ensure that the POD assumptions are still valid

*...without this last step, do you really know if your POD (capability) assumptions still apply?*

# JENTEK's Model-Based Approach (continued)

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- Spatial filtering and data processing to achieve\*:
  - Linear crack response vs. crack size over range of interest
  - Constant variance within accepted bounds over range of interest
- Must include false indication rate with all POD curves and when comparing performance, **# of false alarm opportunities must be the same**

\*the above conditions underlie the properties necessary for POD estimation as reflected in MIL-HNDBK 1823 and can often be achieved with appropriate transformations of signal responses and crack sizes. POD may be estimated without these assumptions being true, but will require methodologies beyond those presented in 1823.

# Model-Based Calibration (with MWM-Arrays)

Sensor in “air”



Shunt Tip



**Air, Shunt Calibration (No Crack Standards) now a U.S. Navy and U.S. Air Force Standard Practice**

**Easy to Replace Cartridges:**



- Sensor
- Shuttle
- Balloons

# First Air Calibration Validation & Verification

1993 Materials Evaluation Paper, Goldfine, (Melcher)

Multivariate Inverse Method using Pre-computed Measurement Grids

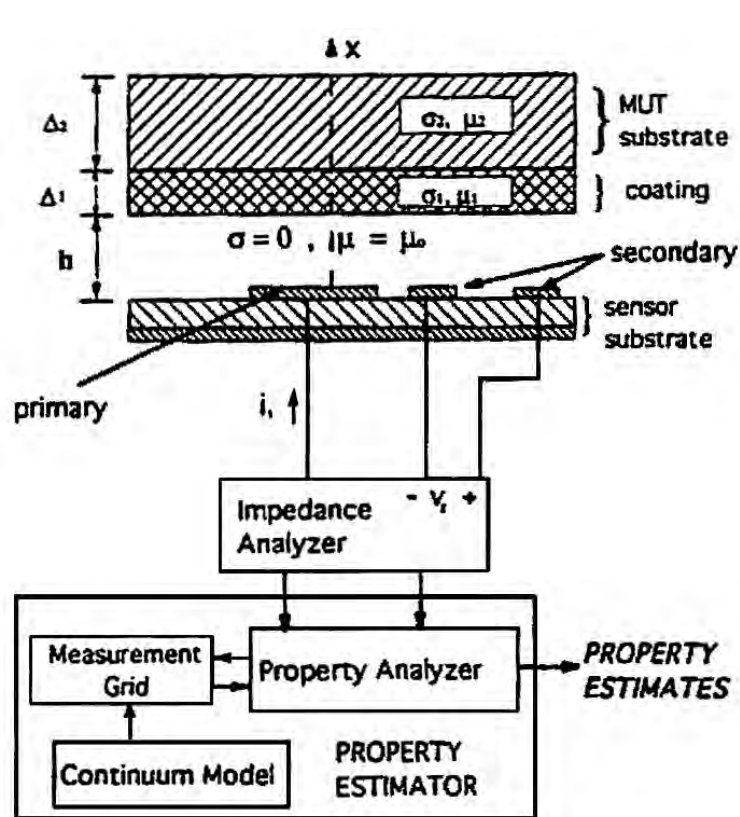


Figure 5—MWM prototype measurement system.

Published, Materials Evaluation 1993

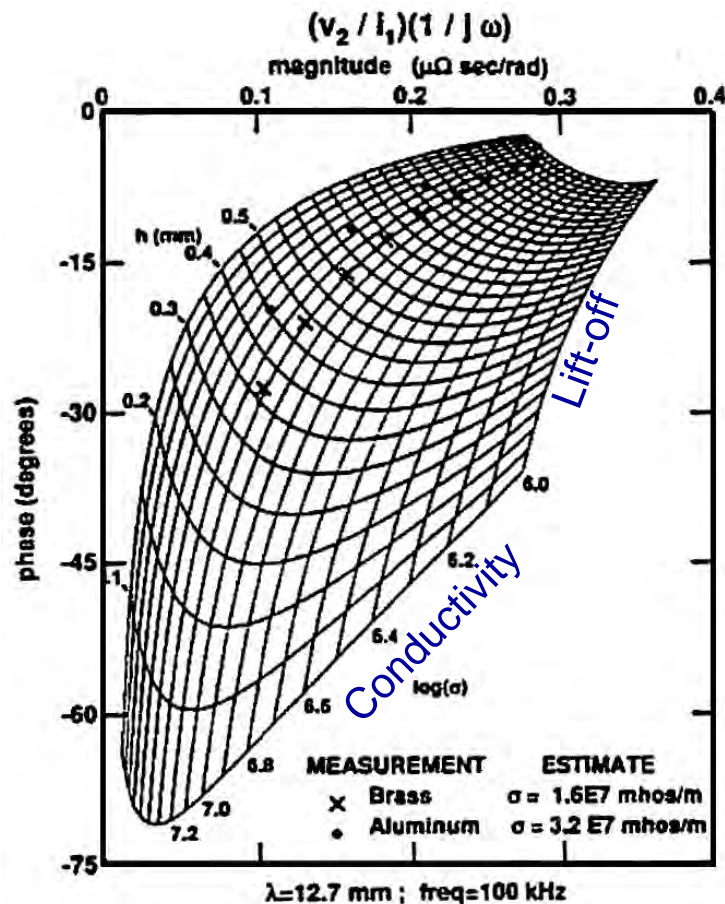
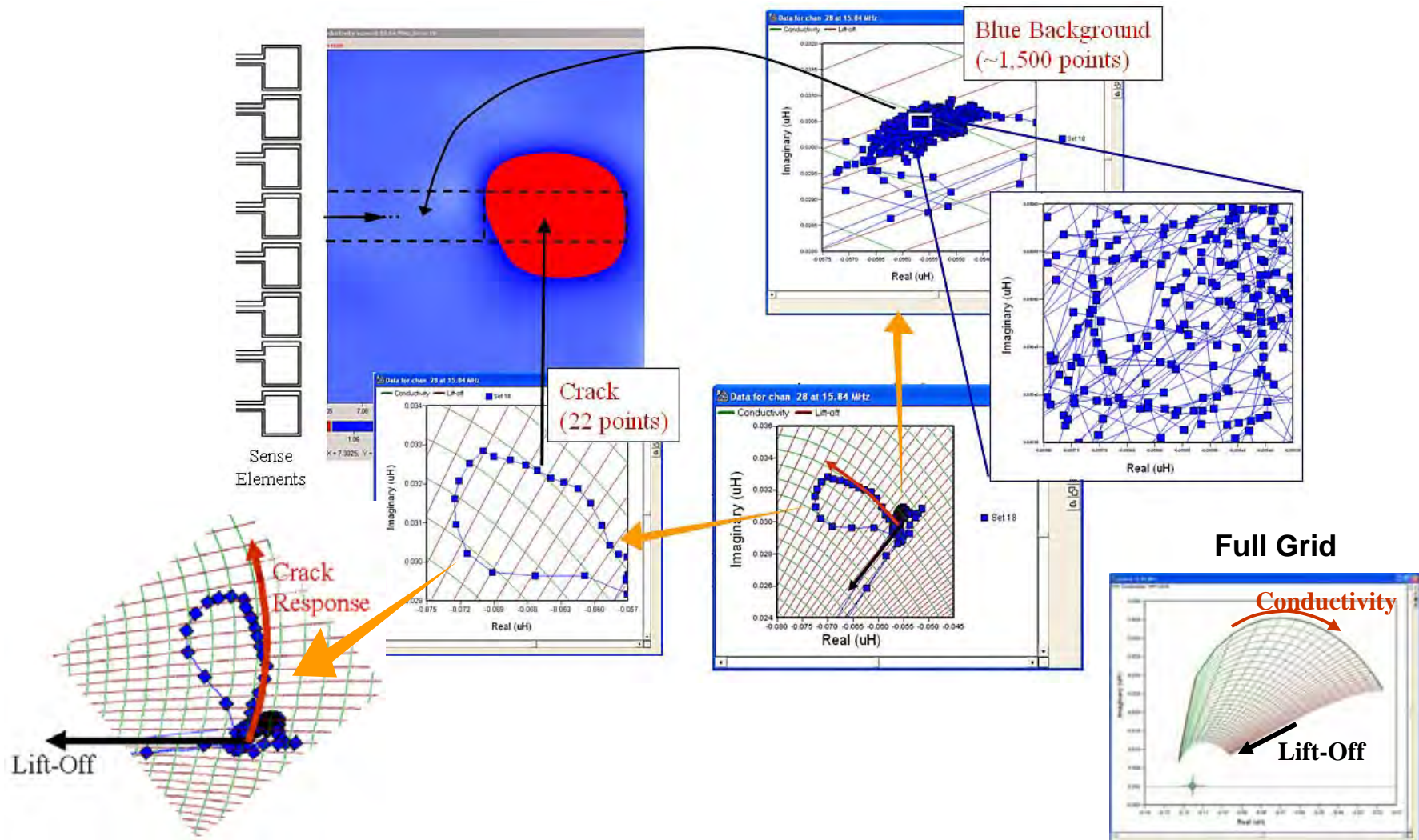


Figure 6—Property estimation grid for independent conductivity,  $\sigma$ , and liftoff,  $h$ , measurement ( $\lambda = 12.7 \text{ mm}$ ,  $f = 158 \text{ KHz}$ ).

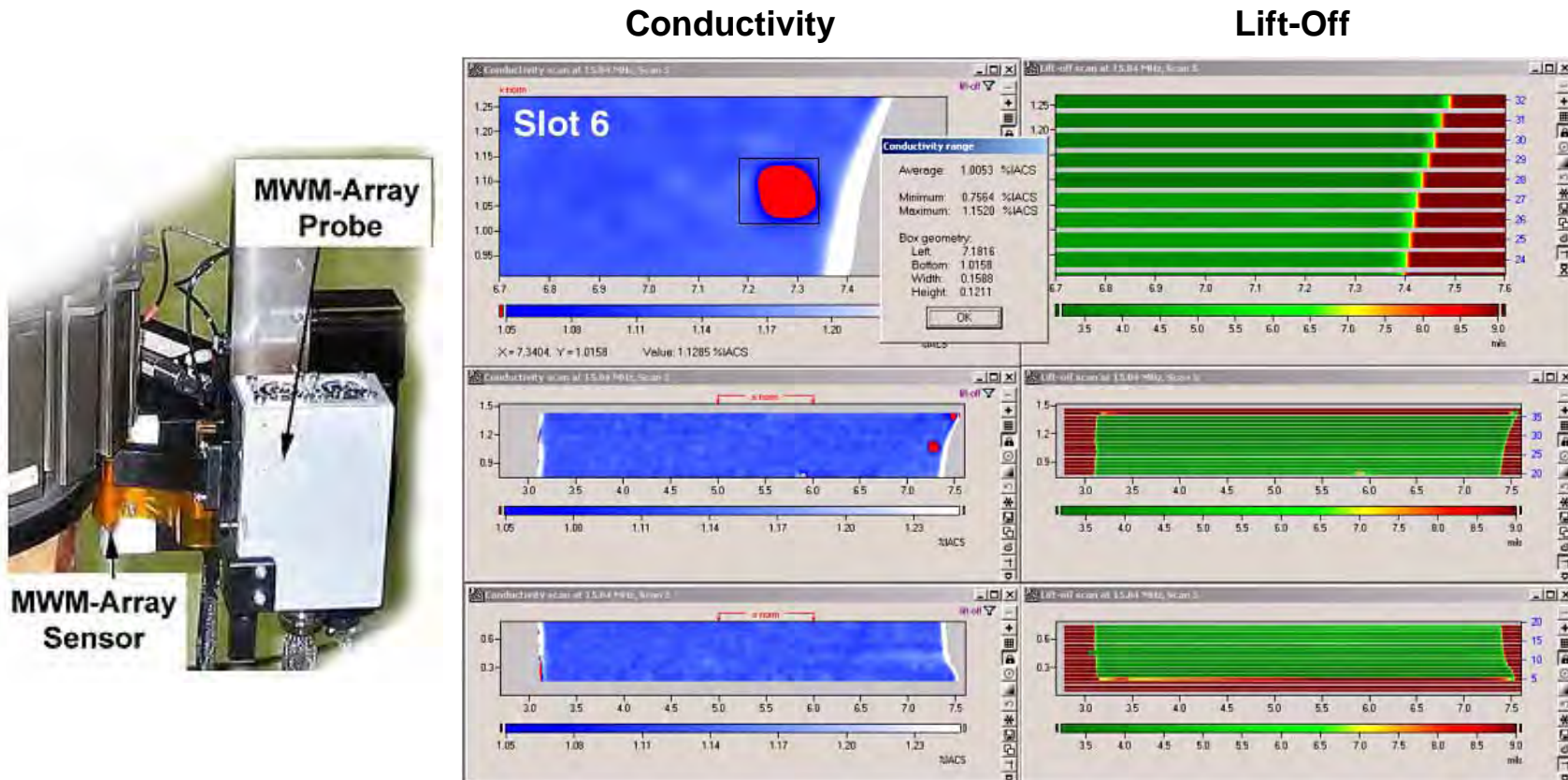
# Model-Based Measurement/Inverse Methods (with MWM-Arrays)

## Rapid Data Processing with Grid Methods and "Air" Calibration



# Model-Based Calibration Verification before, during and after inspections (with MWM-Arrays)

- In use at NAVAIR Depot since April 2005
- Disks with verified cracks detected, several of these verified large and small cracks **not detected by conventional ET and LPT**
- No false indications above threshold after over 7000 inspections



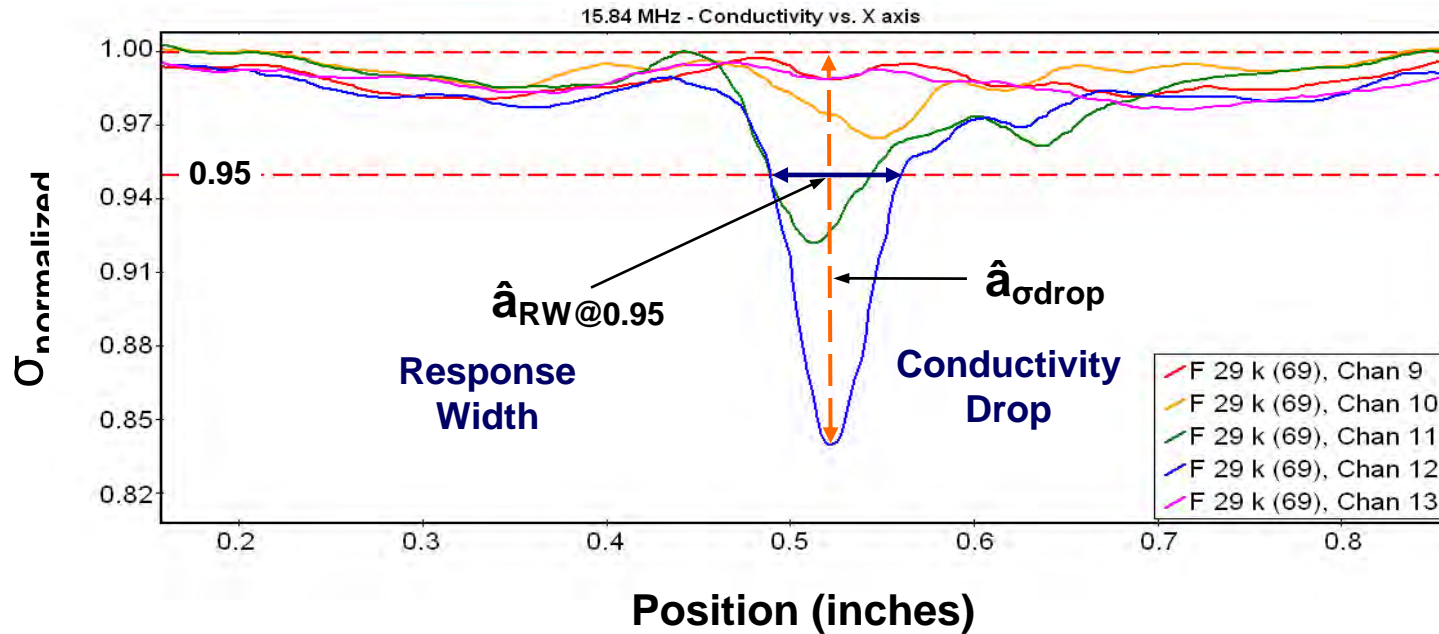


# Definitions for MWM-Array crack response

$\hat{a}_{\sigma\text{drop}}$  VS  $\hat{a}_{\text{RW}@0.95}$



Coupon Fatigue Crack Response (16.9 mils)



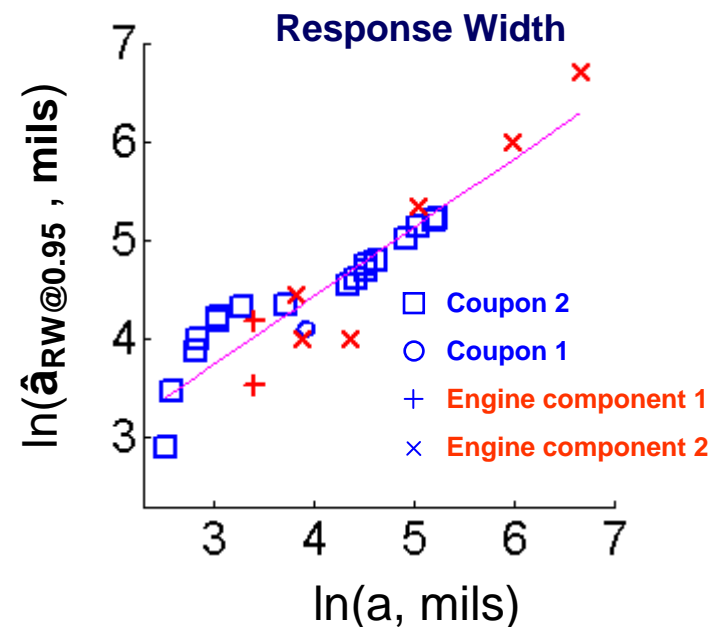
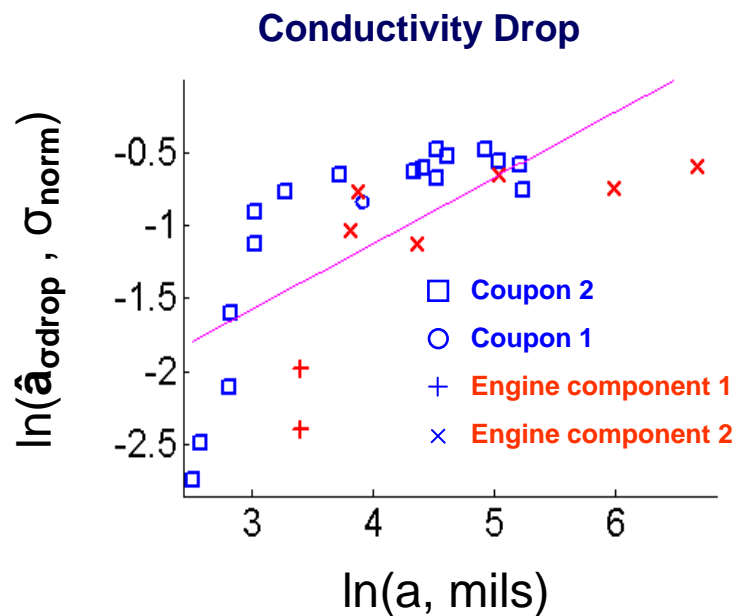
# $\hat{a}$ vs $a$ Plots for Service Parts and Coupons

$\hat{a}_{\sigma\text{drop}}$  VS  $\hat{a}_{\text{RW}@0.95}$



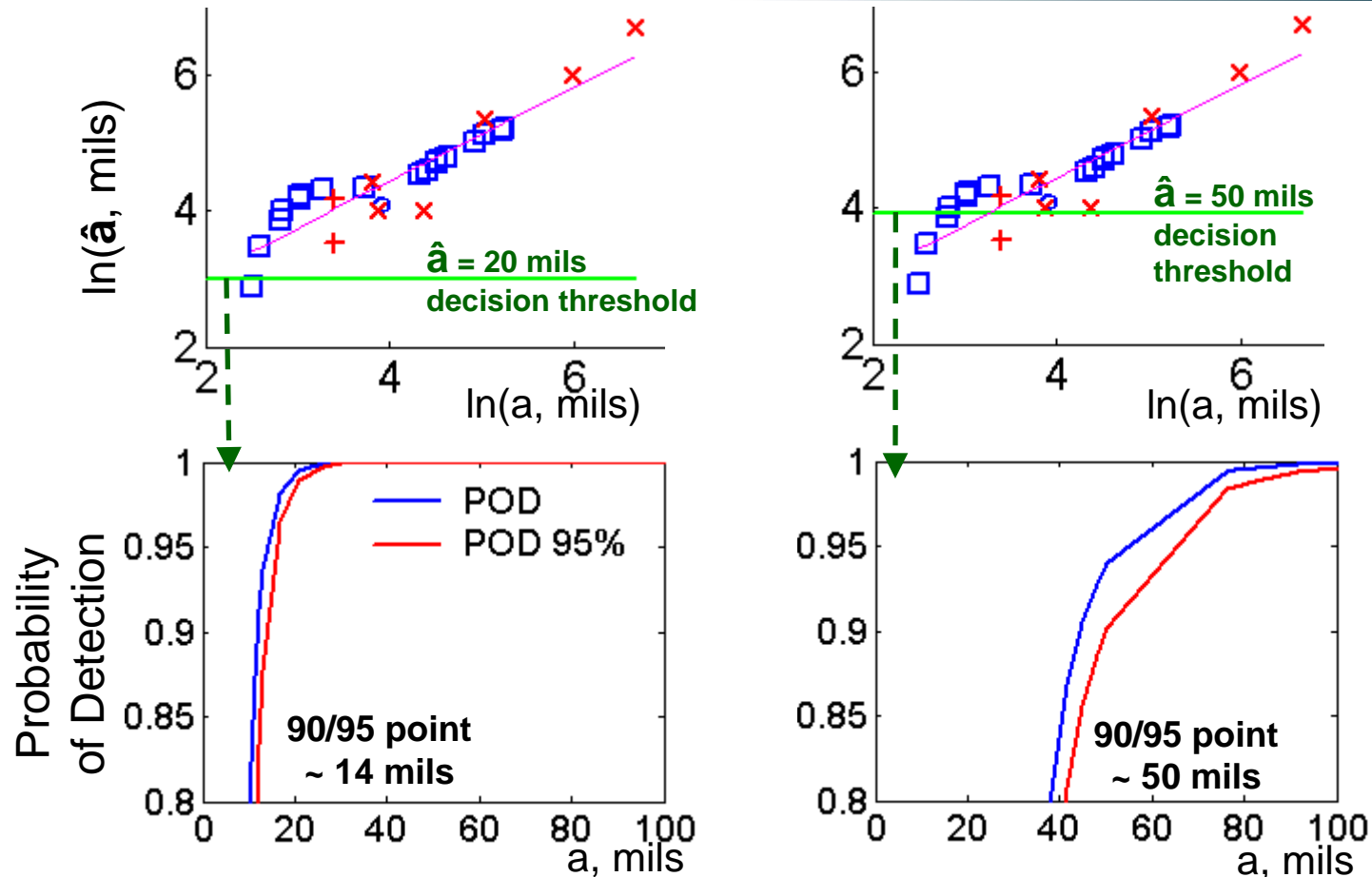
Log( $\hat{a}_{\sigma\text{drop}}$ ) vs log( $a$ ) regression

Log( $\hat{a}_{\text{RW}@0.95}$ ) vs log( $a$ ) regression



**Note: Crack length correlates better with the  $\hat{a}_{\text{RW}}$  (the response width).**

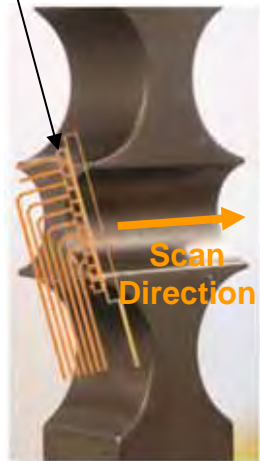
# POD Curves Generated using $\hat{a}_{RW@0.95}$ vs a Data



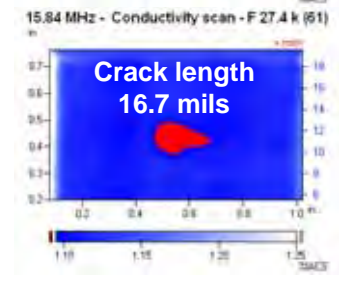
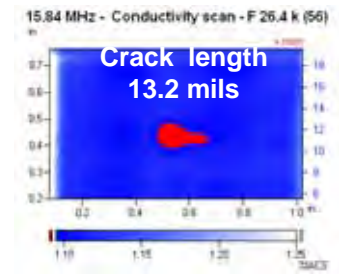
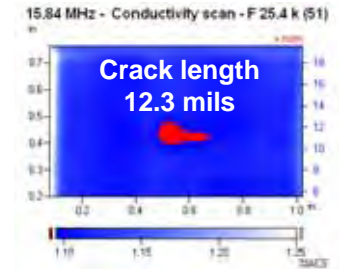
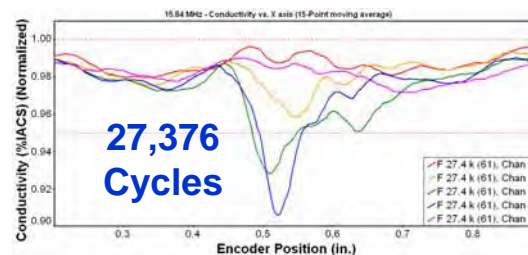
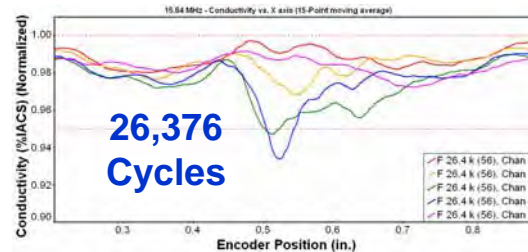
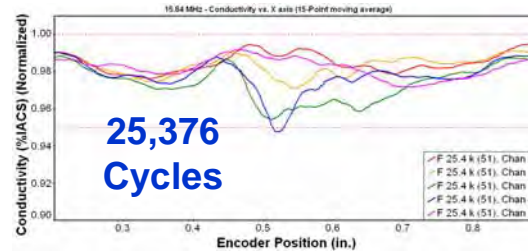
**Note:** Thresholds set above 0.04 inch for the  $\hat{a}_{RW@0.95}$  response result in high confidence ( $\sim 95\%$ ) that the same size crack lengths are detectable with high probability ( $\sim 0.9$ ).

# Automated $\hat{a}$ vs a Data Generation using Multiple Coupons

MWM-Array



Interval	Total Cycles	a, mils
0	0	0
2	6,000	3.9
11	27,376	16.7
12	29,041	16.9
25	41,000	186.6

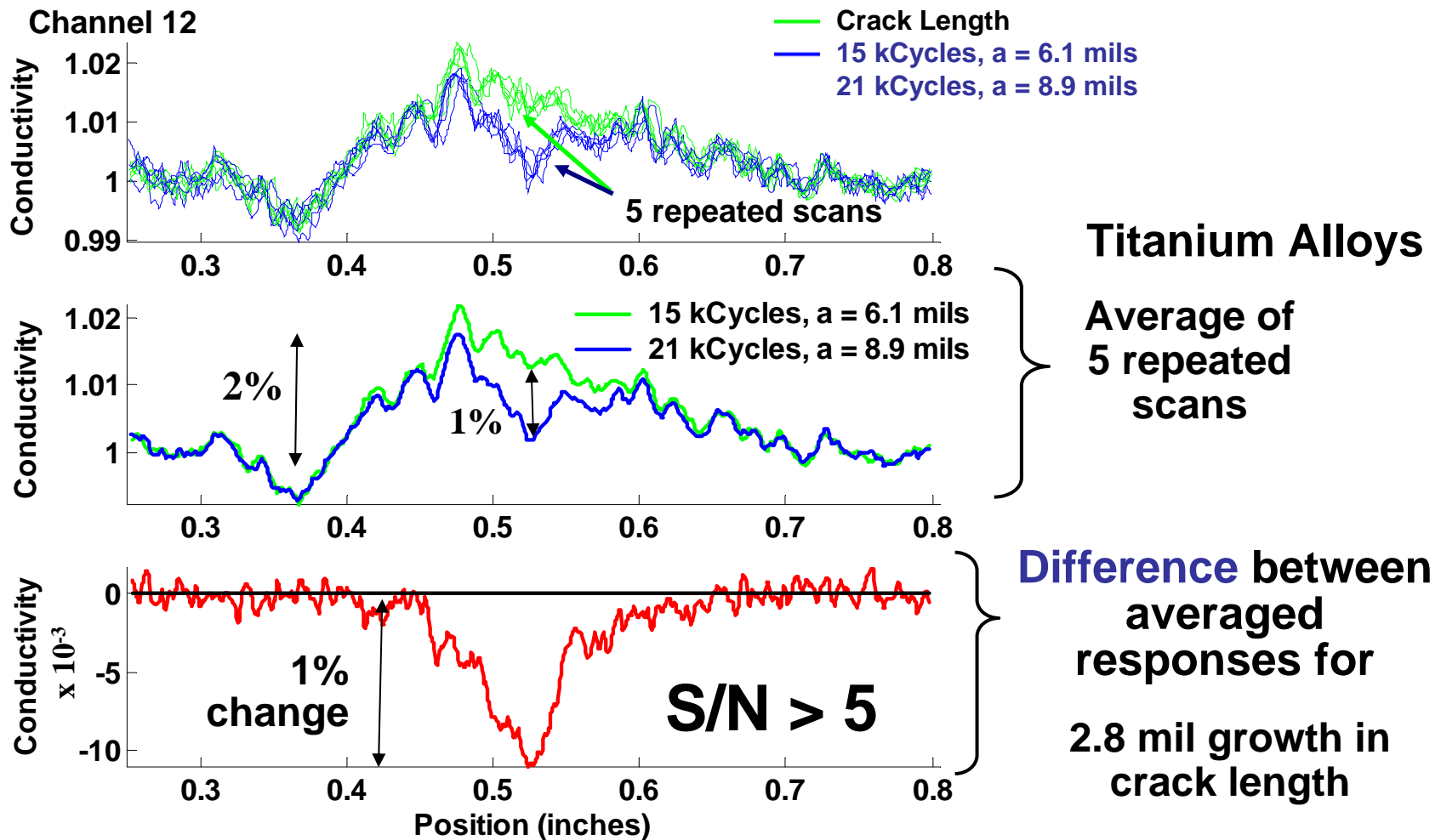


See also: Goldfine, et al, Defense Working Group 2011; Goldfine and Sheiretov, ENDE Conference 2009.

# Difference Imaging or Baseline Subtraction

Improves Signal-to-Noise Levels to Reliably Detect Smaller Cracks

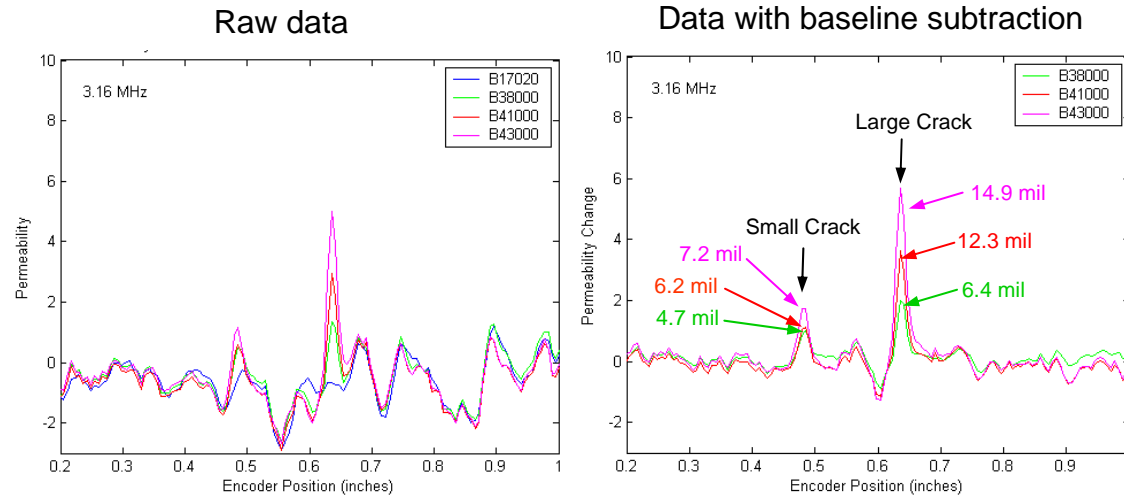
## Titanium



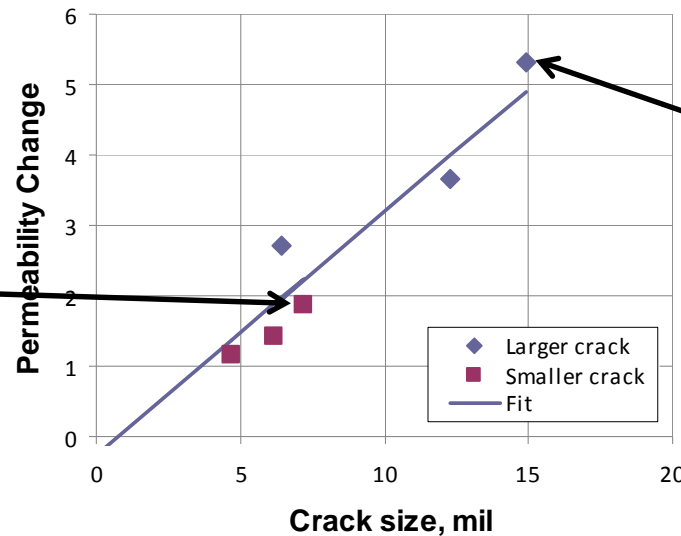
# Difference Imaging or Baseline Subtraction

## Improves Signal-to-Noise Levels to Reliably Detect Smaller Cracks

### A514 Grade B Steel



Crack on front side at 43,000 cycles

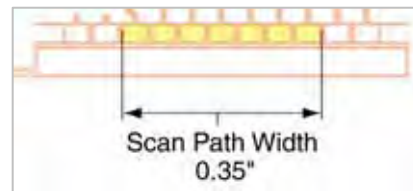


Crack on back side at 43,000 cycles

# Example: Reliability for Bolt Hole Inspection

- C-Scan Imaging using MWM-Arrays
- Detection of Cracks at Edges with edge location correction
- Spatial Filtering for Cracks at Edges

FA43  
Sensor  
Detail



Mandrel Assembly  
with interchangeable  
MWM-Arrays



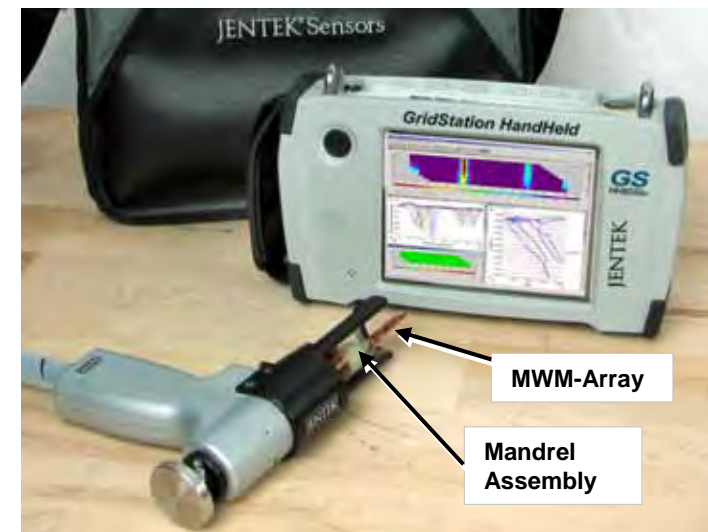
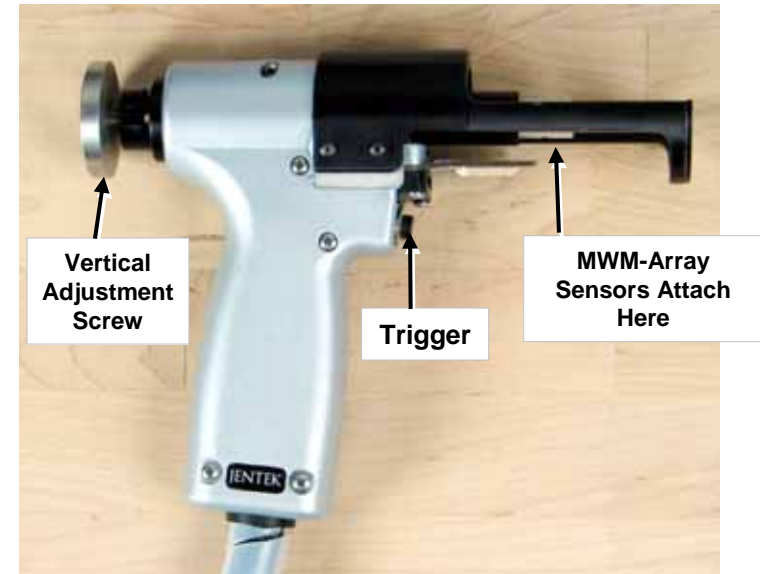
FA166



FA182

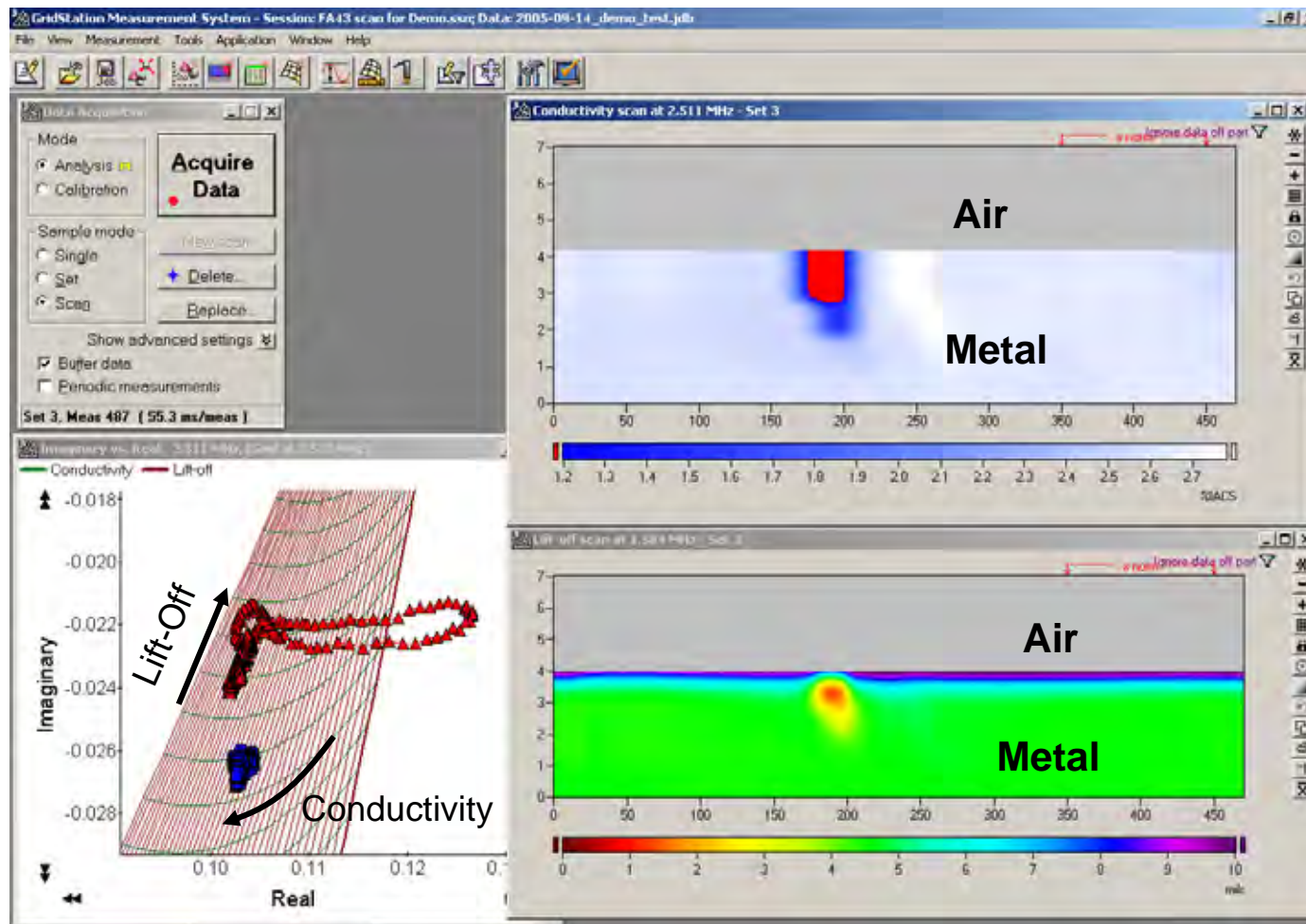


FA43

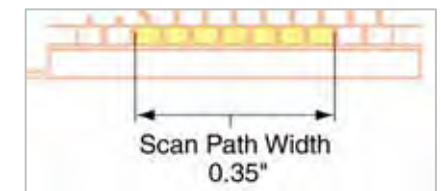


# Correcting for Edges and Other Interferences

## GridStation Conductivity/Lift-Off Images (Unfiltered)



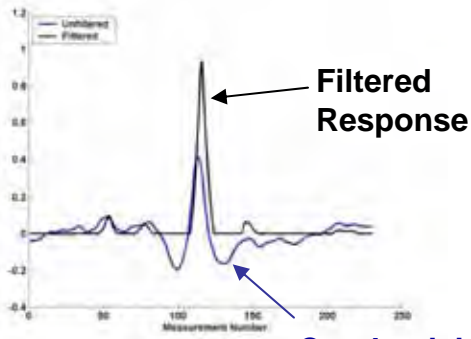
FA43 Sensor Detail



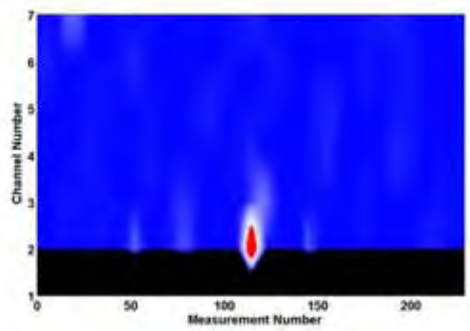
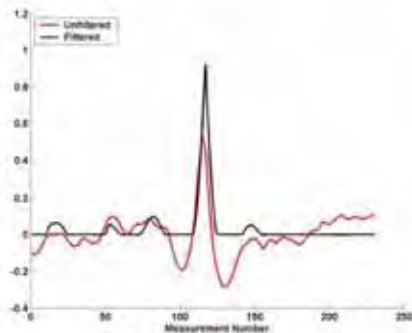
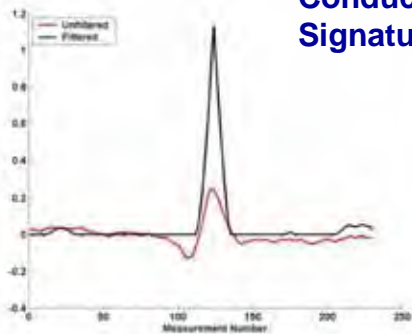


# Detection of Cracks at Edges

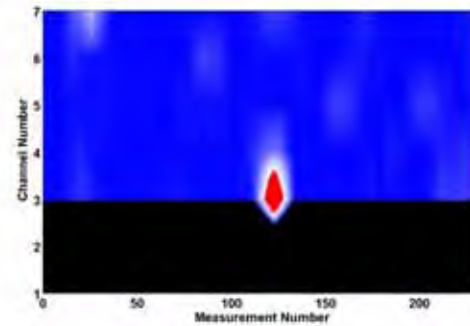
## Edge location correction, and Spatial Filtering, using Signature Libraries



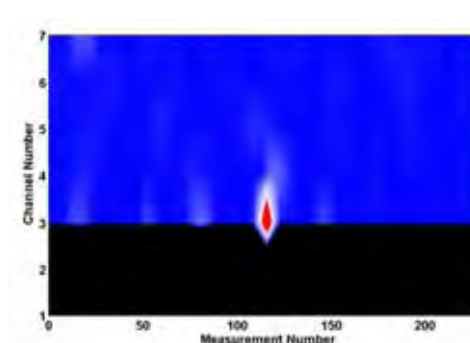
Conductivity Signature



Channel 2, Lift-Off Factor = -0.69



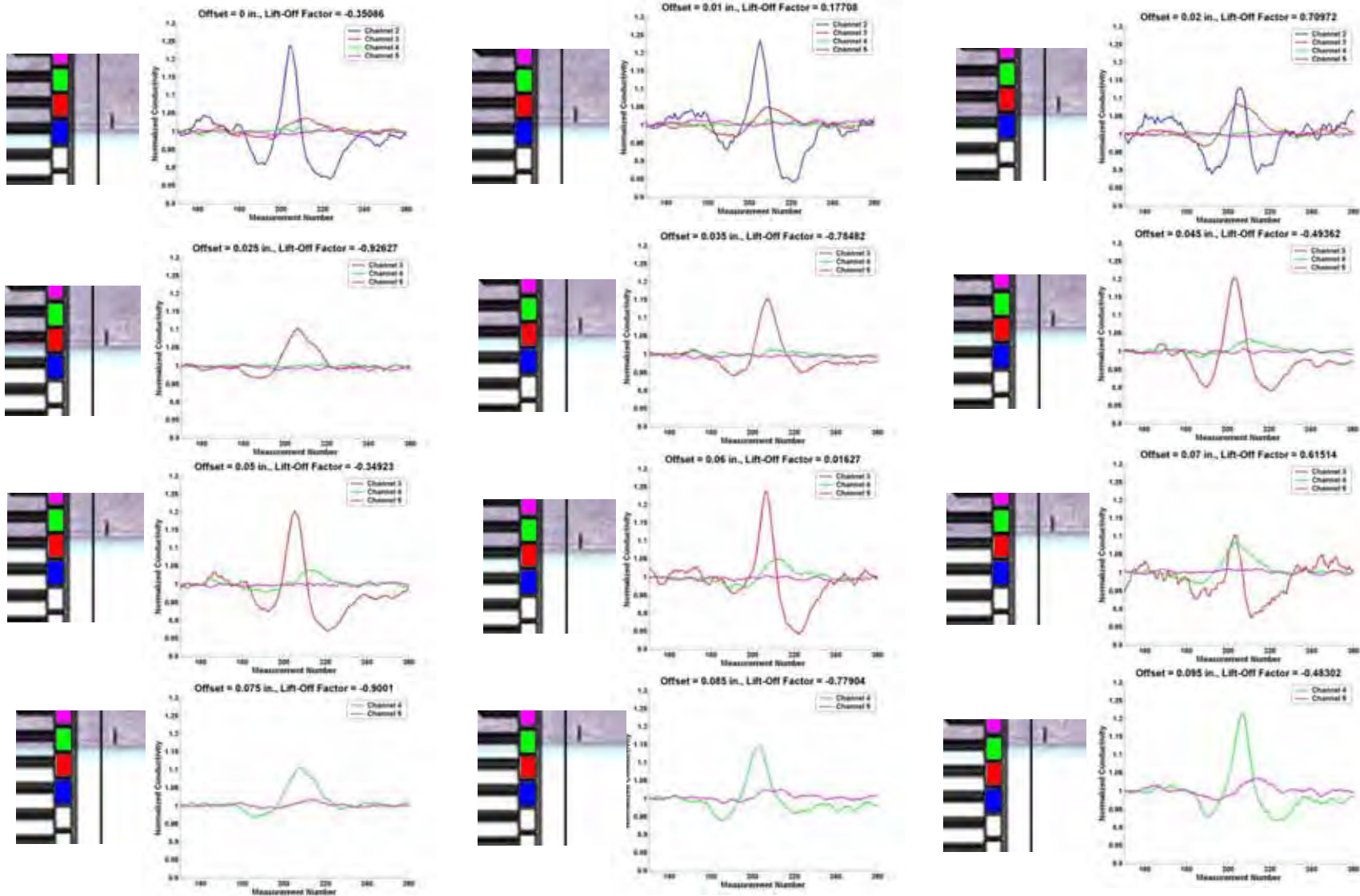
Channel 3, Lift-Off Factor = -0.96



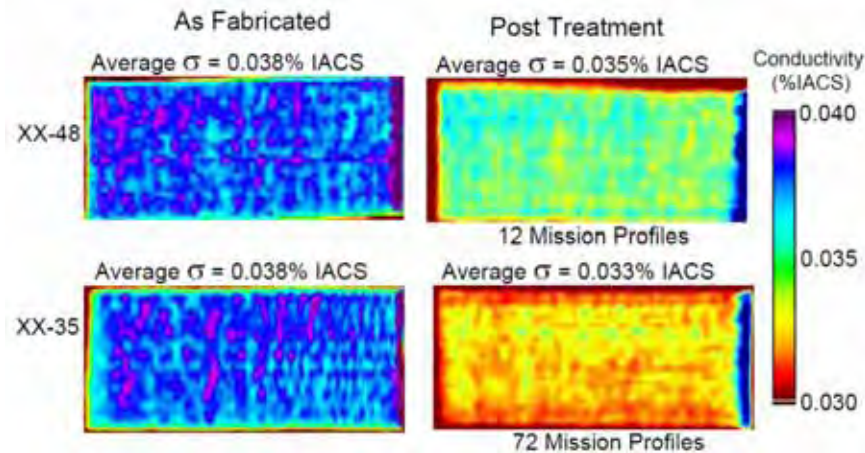
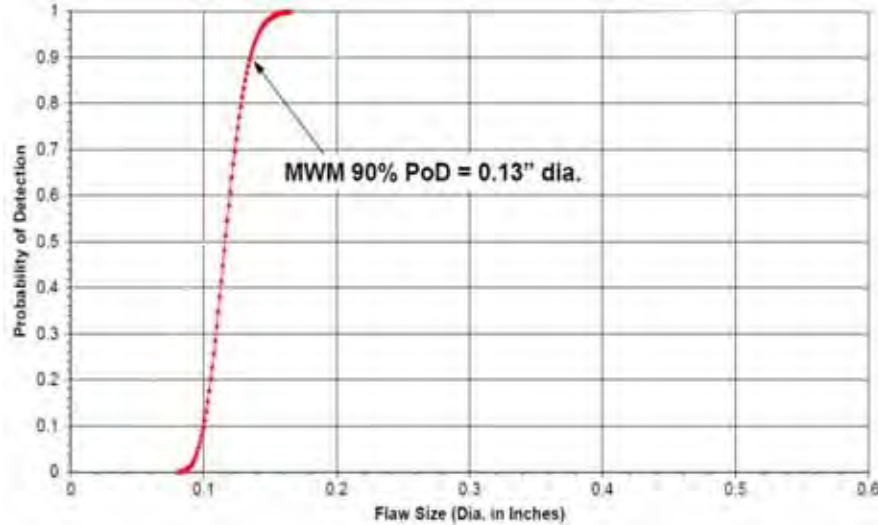
Channel 3, Lift-Off Factor = -0.47



# Signature Library



# Reinforced Carbon-Carbon (RCC) POD




Wincheski, B. Simpson, J., "Application of Eddy Current Techniques for Orbiter Reinforced Carbon-Carbon Structural Health Monitoring," CP820, Review of Quantitative Nondestructive Evaluation, Volume 25, PART B, pages 1082-1089, ed. by D.O. Thompson and D.E. Chimenti.

# Engine Component

## OEM & FAA - Approved Engine Component NDT with MWM-Arrays

**AE SERIES PROPULSION SYSTEM**  
Service Bulletin Index



**Rolls-Royce**

LIST OF AE 3007A SERIES SERVICE BULLETINS

SB No.	Rev No.	Title	Compliance Category	Date	Models Affected	Module or ATA Locator
AE 3007A-72-386		See AE 3007A-A-72-386				
AE 3007A-72-388	1	Engine - 8th- thru 13th-Stage Compressor Wheel Knife Edge Seals - Jentek Eddy Current Inspection	8	09-May-11	7A, 7A1/1, 7A1/3, 7A1, 7A1E, 7A1P, 7A2, 7A3	72-37-00

***“Technical aspects of the method are FAA approved.”***



MWM-Array  
FA43 Sensor

# Reliability for Permanently Installed Eddy Current Sensors

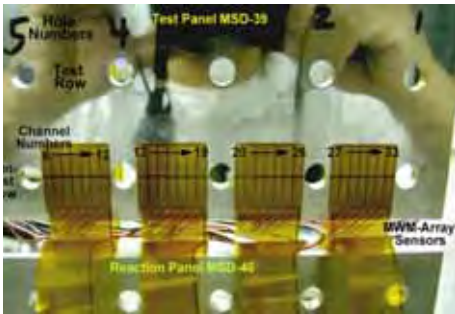
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- Embedded and Surface Mounted
- Linear MWM-Arrays and MWM-Rosettes
- Continuous Monitoring vs. Data Recording on the Ground Only

# Example Linear & Integrated Solutions

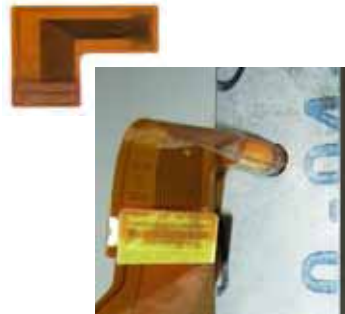
## Example Linear MWM-Arrays

MWM-Array FA47



MWM-Array FA75

MWM-Array FA73

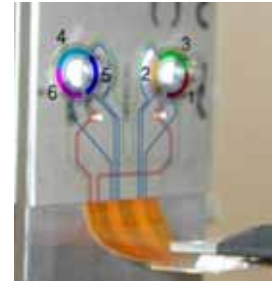
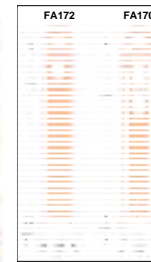
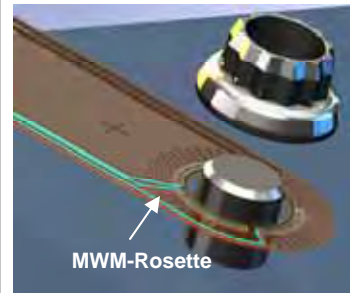


MWM-Array FA120

MWM-Array FA65



## Example MWM-Rosettes & Integrated Solutions

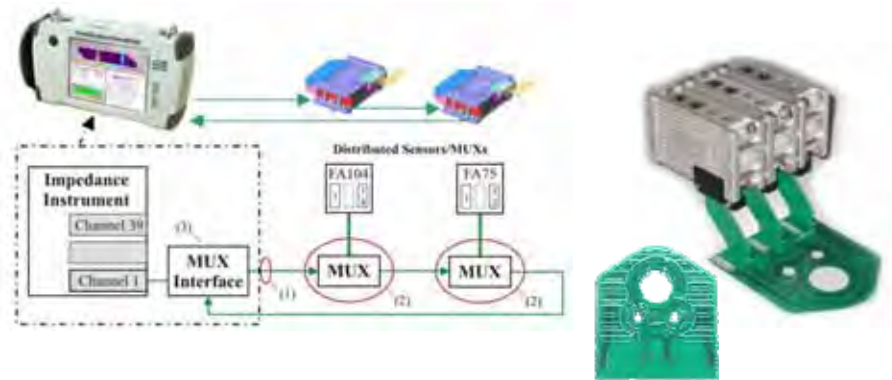


MWM-Arrays FA138, FA140

MWM-Array FA80

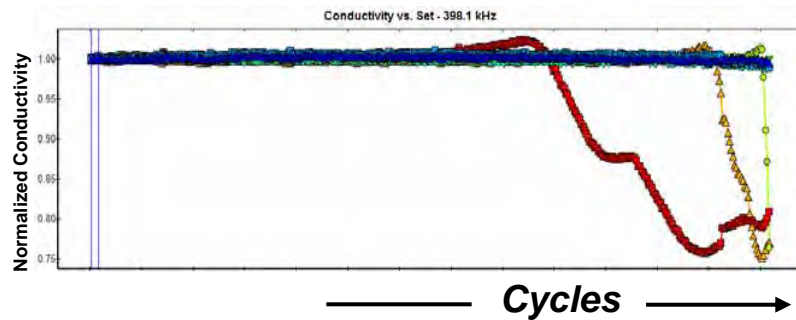
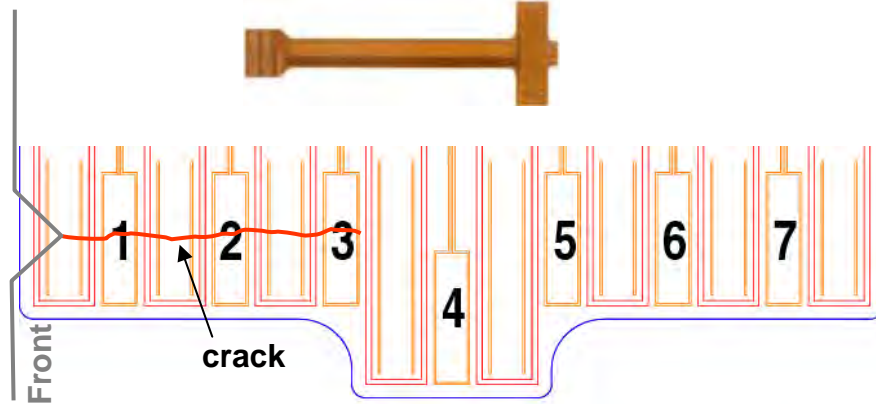


## System and MWM-Array Sensor Mux Network

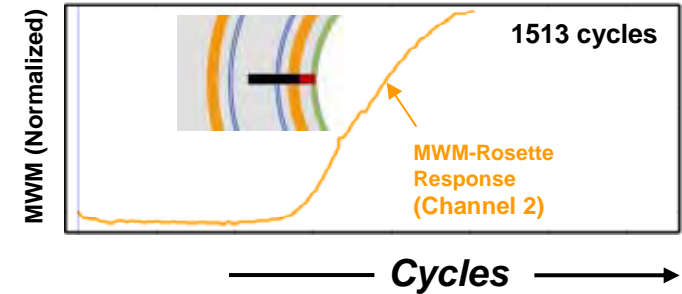
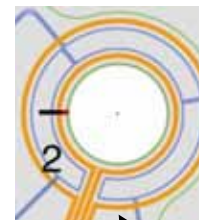
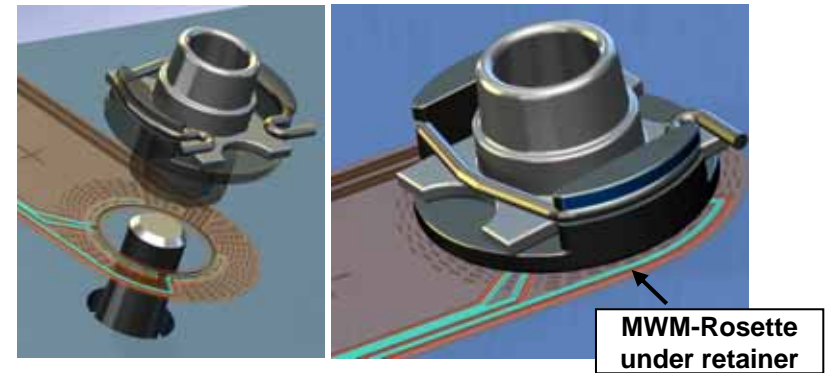


## Linear MWM-Arrays

FA65 MWM-Array



## MWM-Rosettes



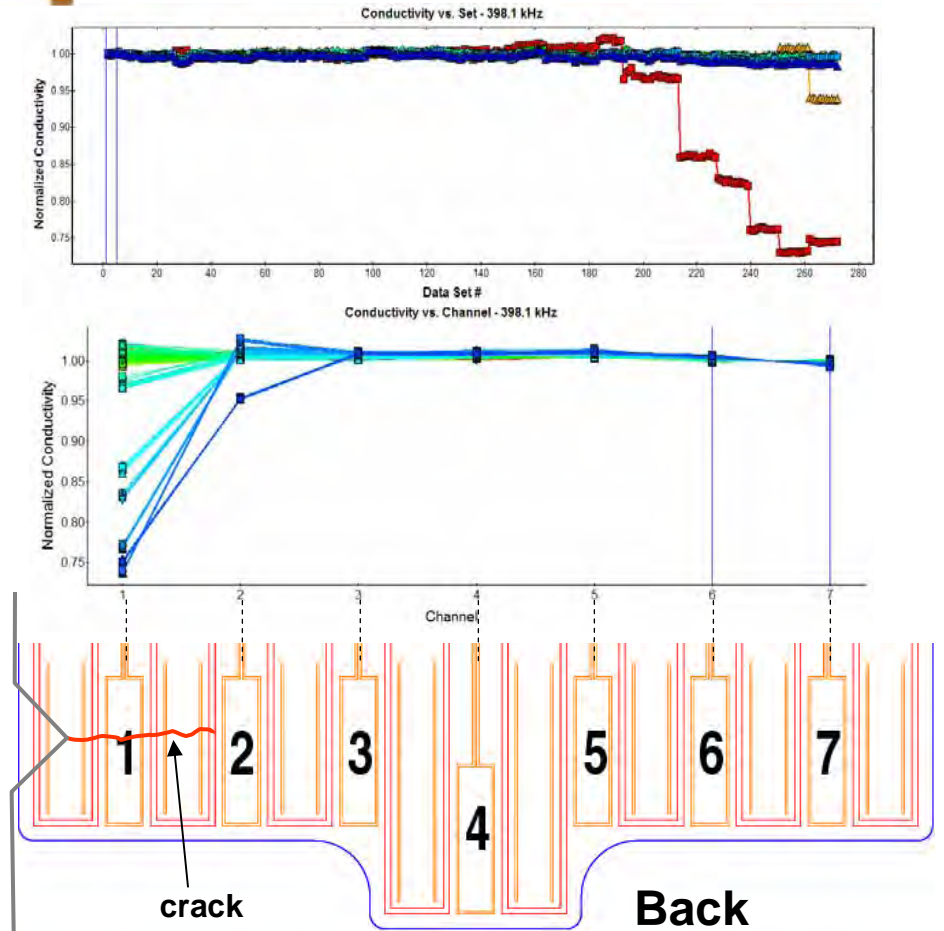
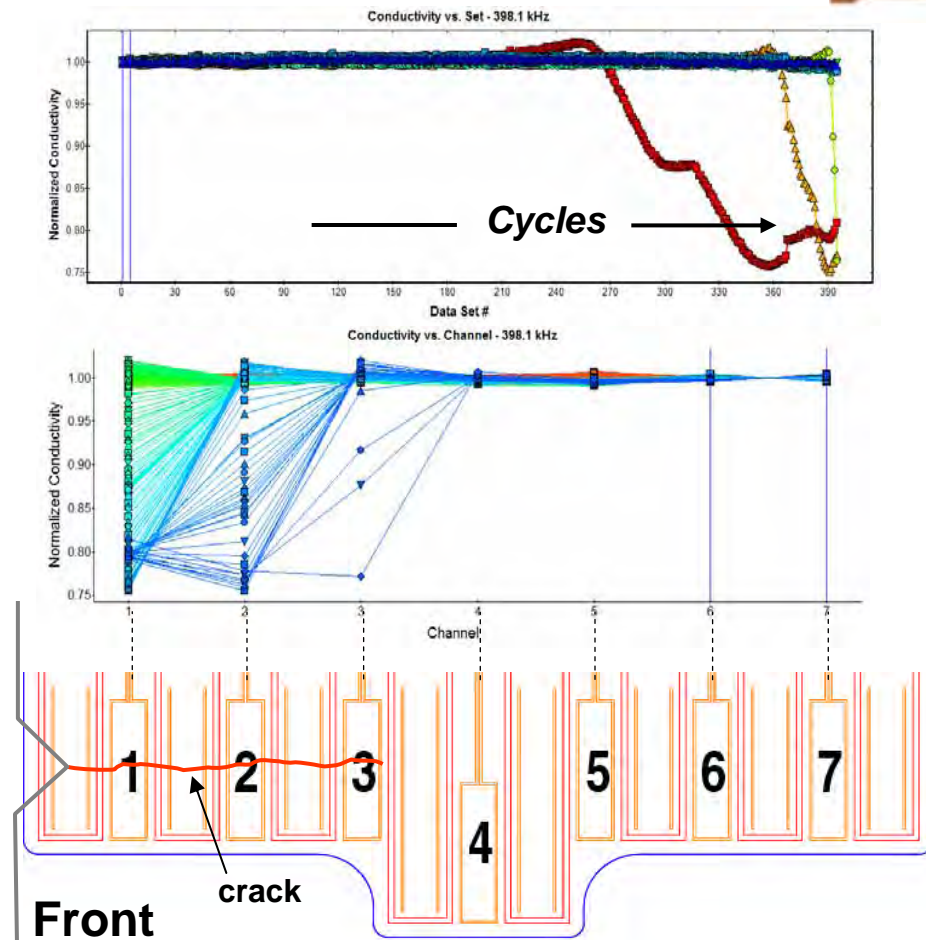
# Linear MWM-Arrays & MWM-Rosettes can provide either continuous or scheduled inspection during fatigue or at rest

Continuous monitoring

FA65 MWM-Array



Scheduled inspections to simulate on-aircraft use



See also "Numerous Embedded Inductive and Capacitive Sensors for Corrosion & Fatigue," Aircraft Airworthiness & Sustainment (AA&S) Conference, Austin, TX, Presented May 2010.

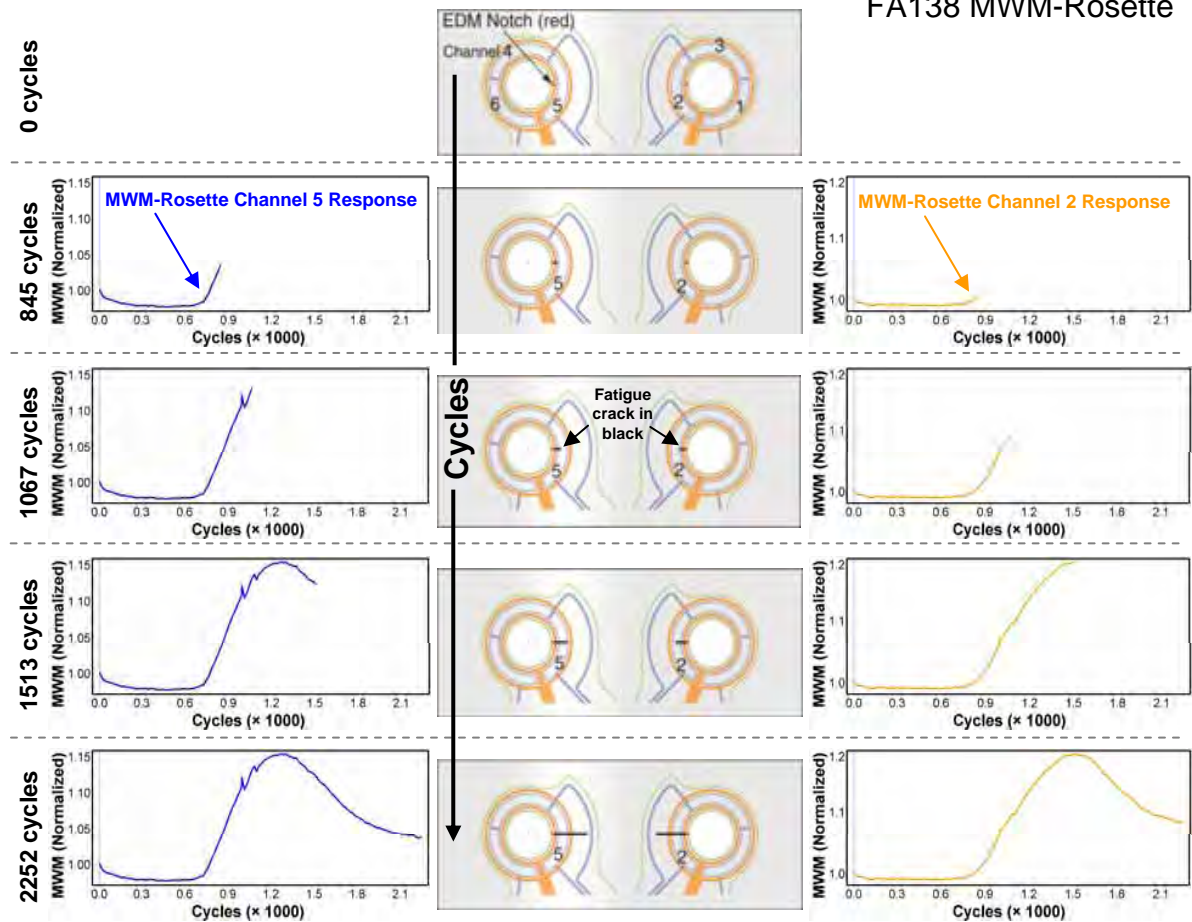
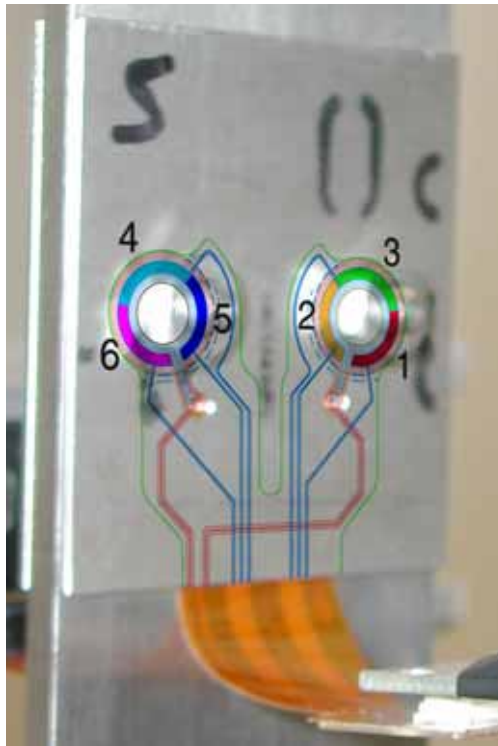


# POD Data Generation MWM-Rosettes

- MWM-Rosette (FA138) response monitored during fatigue test
- Determining actual crack sizes during testing
- Run multiple tests (e.g., 3-7 coupons)



FA138 MWM-Rosette



# $\hat{a}$ vs $a$ Response Model in Linear Range

---

***MWM Crack Length Estimate =  $\hat{a}$***

$$\hat{a} = 1 + (b_0 + \varepsilon_s) a + \varepsilon_r$$

$b_0$  = Average **slope** for MWM response versus flaw size,

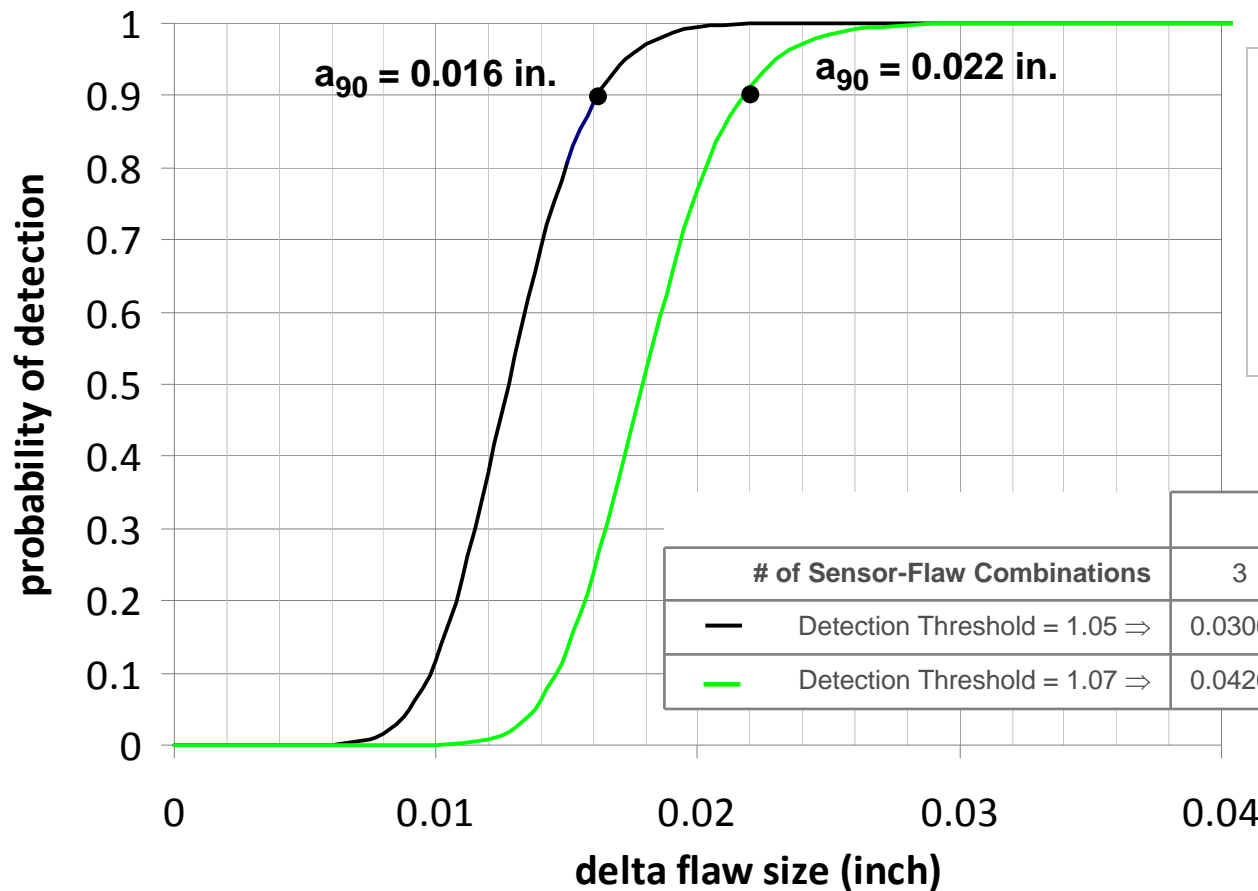
$\varepsilon_s$  = Gaussian random variable with mean 0 and standard deviation  $\sigma_s$  (**sensor sensitivity, slope, variation**)

$a$  = **Actual Flaw length**

$\varepsilon_r$  = Gaussian random variable with mean 0 and standard deviation  $\sigma_r$  (**“glitches” from interrupted testing and other sources**)

# First POD Curves for Embedded Eddy Current Sensors using Phase I Coupon Data

- Phase I data limited to 2 flaws  
 $b_0$  est. = 3.920,  $\sigma_s$  est. = 0.400, and  $\sigma_r$  est. = 0.0082



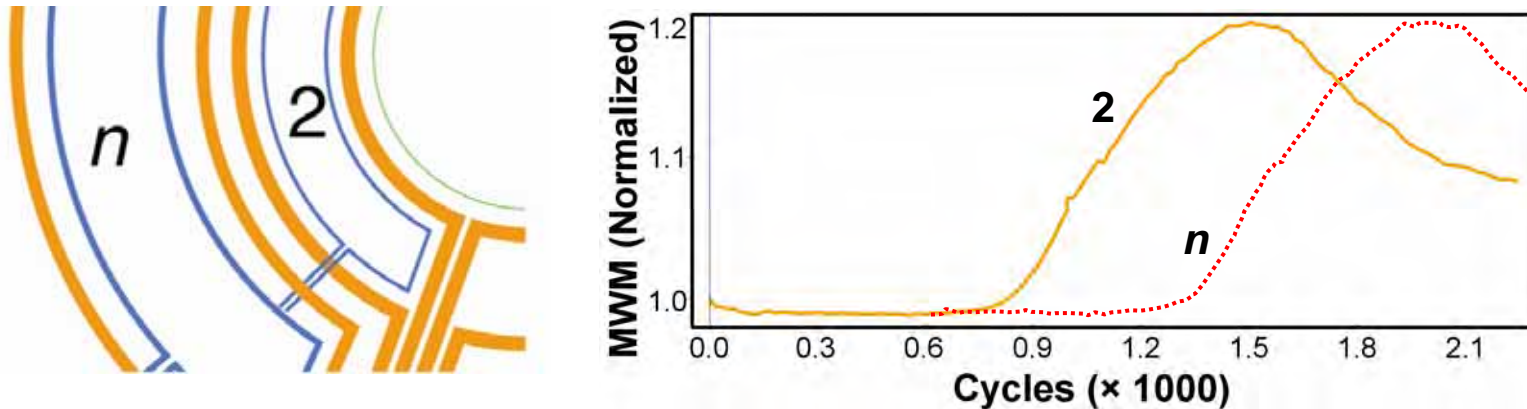
**How many coupon tests do we need?**



		$a_{90/95}$		
# of Sensor-Flaw Combinations		3	5	10
—	Detection Threshold = 1.05 $\Rightarrow$	0.0306	0.0193	0.0173
—	Detection Threshold = 1.07 $\Rightarrow$	0.0426	0.0265	0.0236

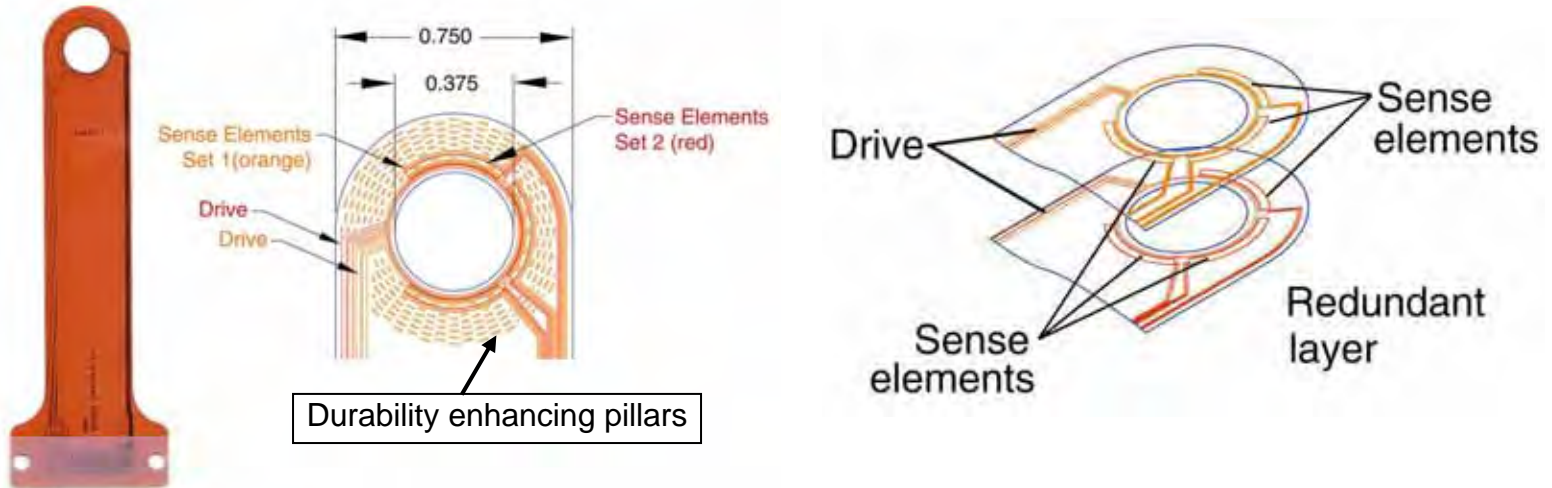
# To Improve POD Curve Estimation

- We can add another channel to expand linear range



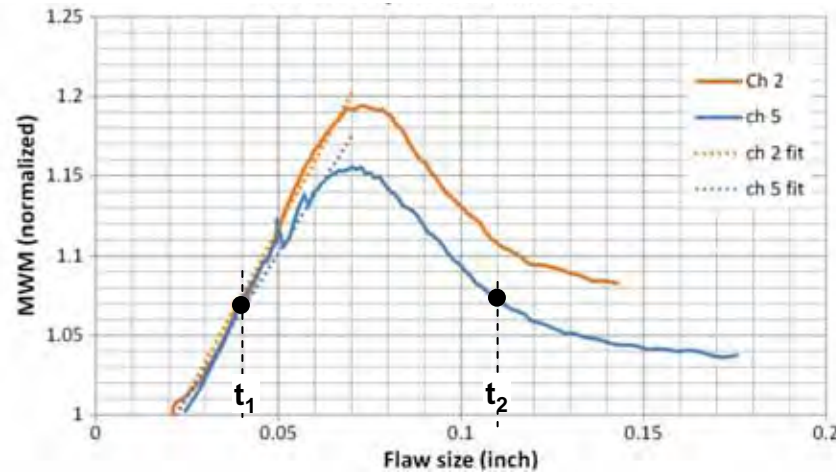
- We can add redundant channels to improve noise suppression

MWM<sup>®</sup>-Rosette FA158

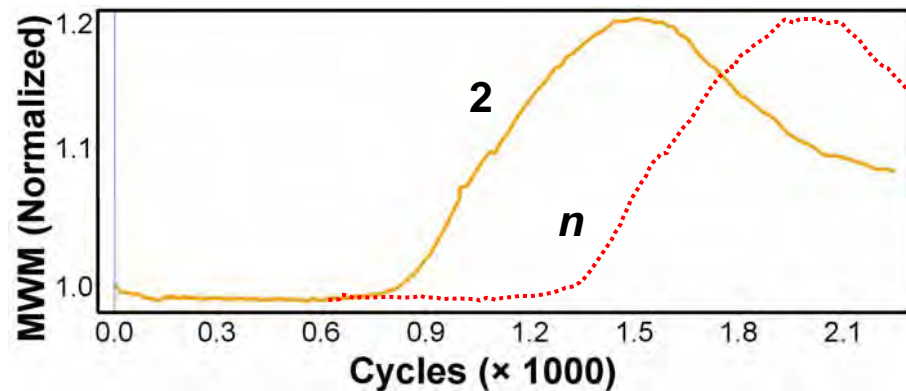
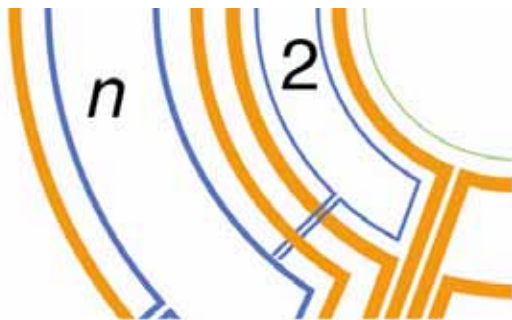


# Caveats and Further Development

- POD form is valid only if able to determine which regime of the curve a single measurement occurs



- Can enable determination of regime with enhanced sensor design (e.g., add a channel)



# JENTEK's Model-Based Approach is

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- **Model-Based** Calibration and Measurement Methods
- Calibration Verification on Crack Standards or actual service hardware with verified defects, if available
- **Calibration verification at each inspection location**, to ensure that the POD assumptions are still valid

*...without this last step, do you really know if your POD (capability) assumptions still apply?*

***Is POI  $\cong$  1.0 ?***

***Remember:***

$$\text{Pr}(\text{detect}) = \text{POD} * \text{POI} + \text{unquantified POD} * (1 - \text{POI})$$

# Questions?

